

# Development of A THz FTS for Site Testing at Dome A









Sheng-Cai Shi<sup>1</sup>

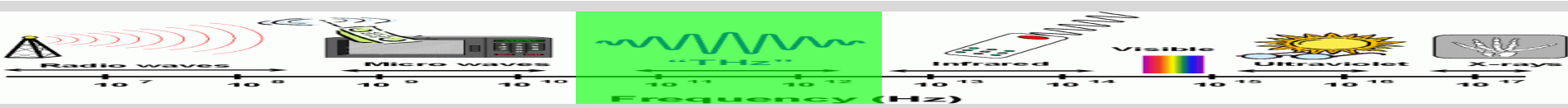
S. Paine<sup>2</sup>, Q.J. Yao<sup>1</sup>, X.X. Li<sup>1</sup>, X.G. Zhang<sup>1</sup>, Z.H. Lin<sup>1</sup>, H.  
Matsuo<sup>3</sup>, J. Yang<sup>1</sup>, Q.Z. Zhang<sup>2</sup>

*A collaboration between PMO<sup>1</sup> and CFA<sup>2</sup>*

# OUTLINE

-  **Introduction**
-  **Design of the System**
-  **Sensitivity & Calibration**
-  **Control/DAQ System & Software**
-  **Timeline for FTS Development**
-  **Conclusion**

# Why do THz Astronomy?



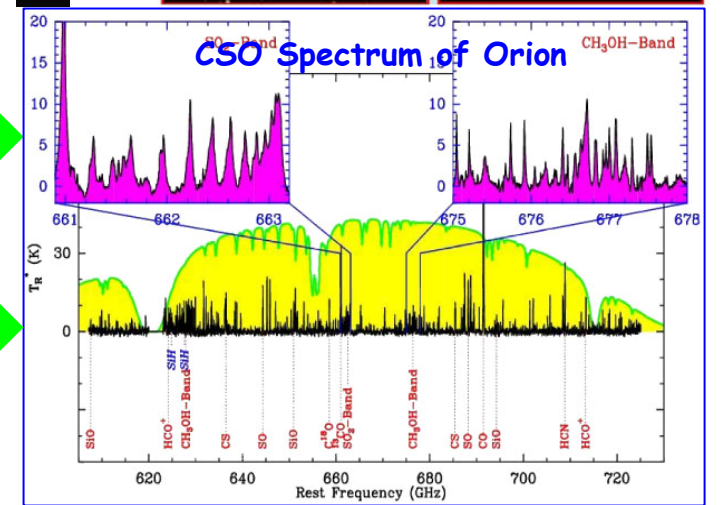
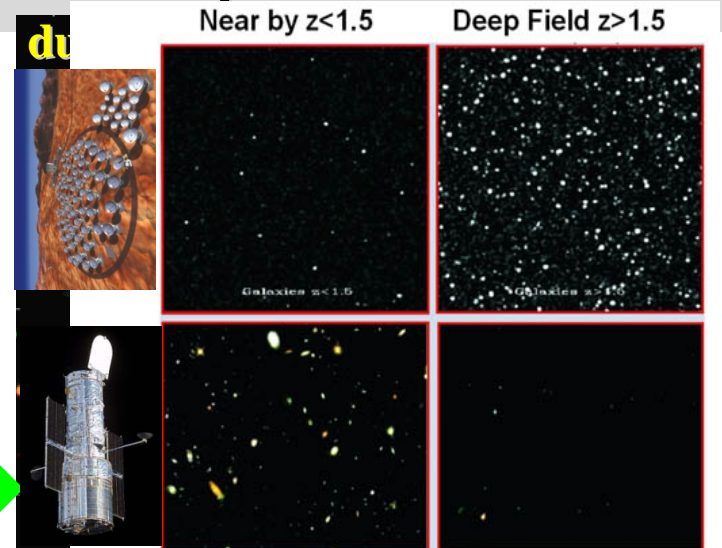
electronics

Size  $\sim \lambda$

$h\nu \sim k_B T$

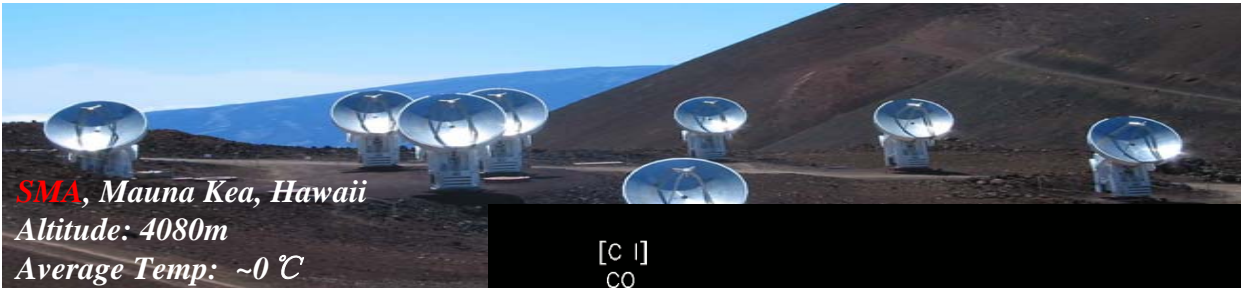
photonics

- ~50% photon energy after CMB
- cold (10K~1THz) objects in formation
- early distant objects (dust reemission and red shift)
- deep inside (seeing through) of objects, optically thin @ THz
- rich molecular rotation lines

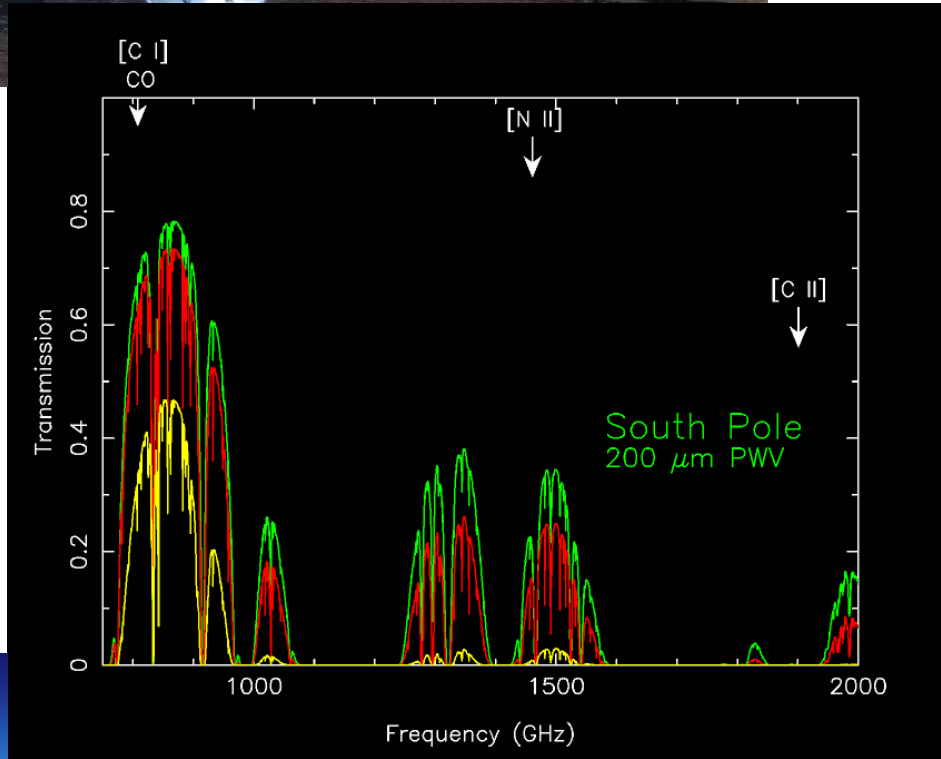




# Good THz Sites on the Earth



**SMA, Mauna Kea, Hawaii**  
Altitude: 4080m  
Average Temp:  $\sim 0^\circ\text{C}$



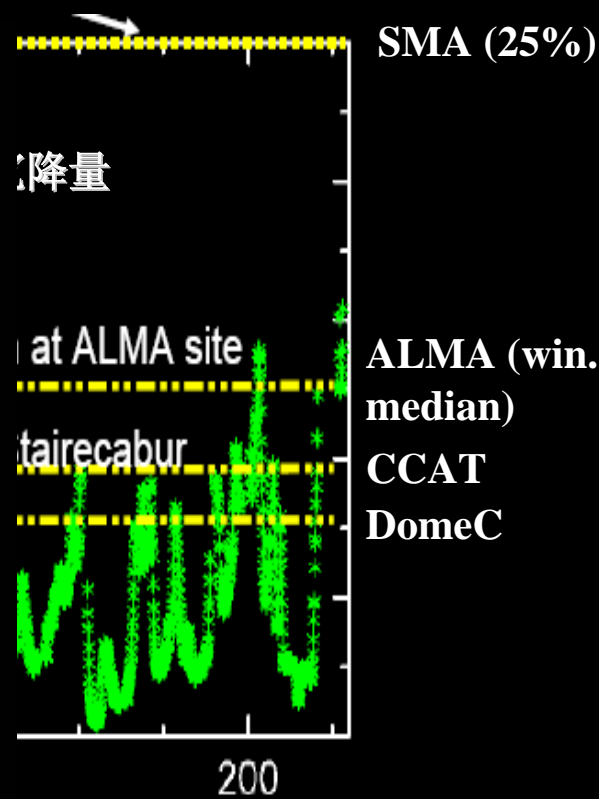
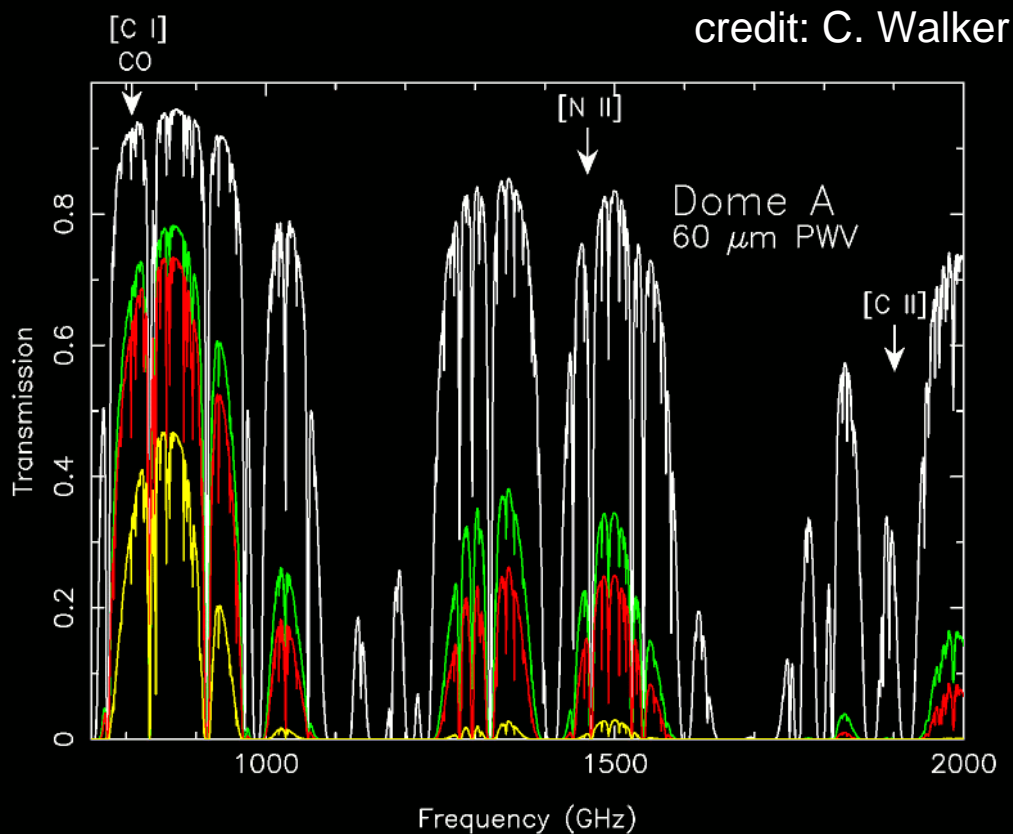
**ALMA, Chajnantor, Chile**  
Altitude: 5100m  
Average Temp:  $-2^\circ\text{C}$



**SPT, South Pole**  
Altitude: 2835m  
Average Temp:  $-49^\circ\text{C}$

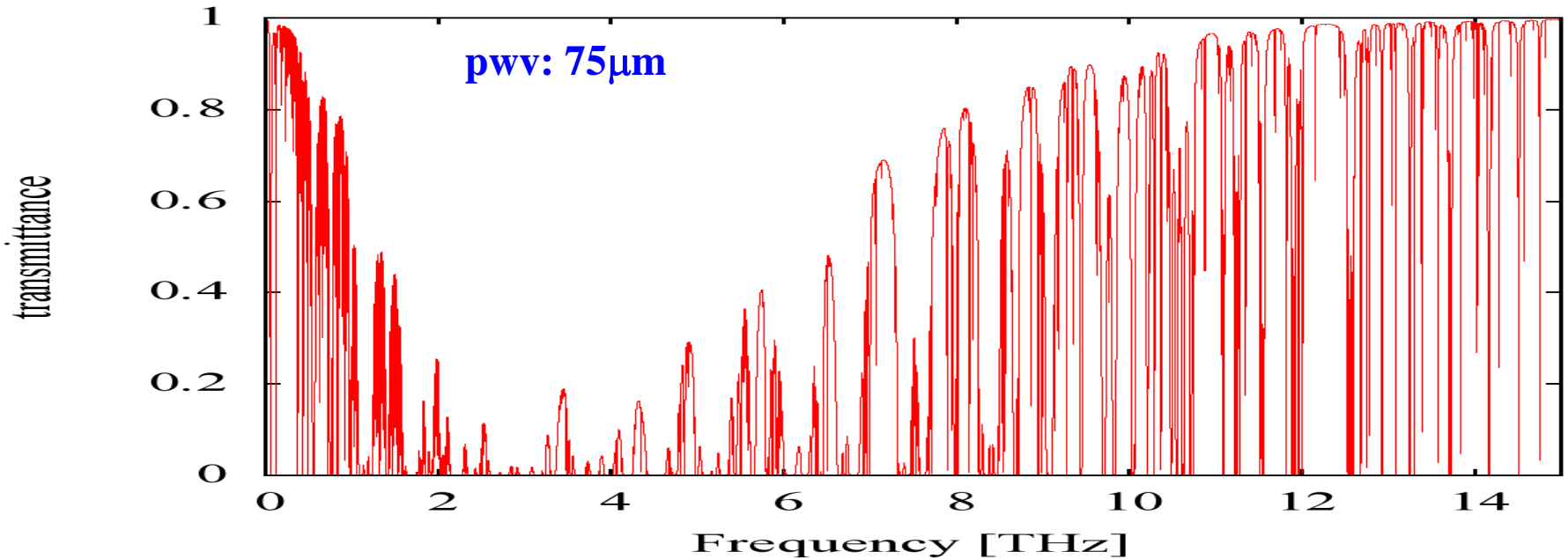
credit: C. Walker

# Dome A: the Best THz Site on the Earth?



# FTS vs Radiometer

	Band-width	Measurement Method	Temp Calibration	Accuracy	Freq Resolution	Other Freq
<b>Radiometer</b>	single freq	tipping with $z$ , $P_{IF}(z) \rightarrow \tau_0$	not needed	good with small $\tau_0$	high	AM needed
<b>FTS</b>	large	$T_{sky}(\nu)$ at fixed $z \rightarrow \tau_z(\nu)$	needed	reliable for $\tau$ up to 5	low	AM not needed



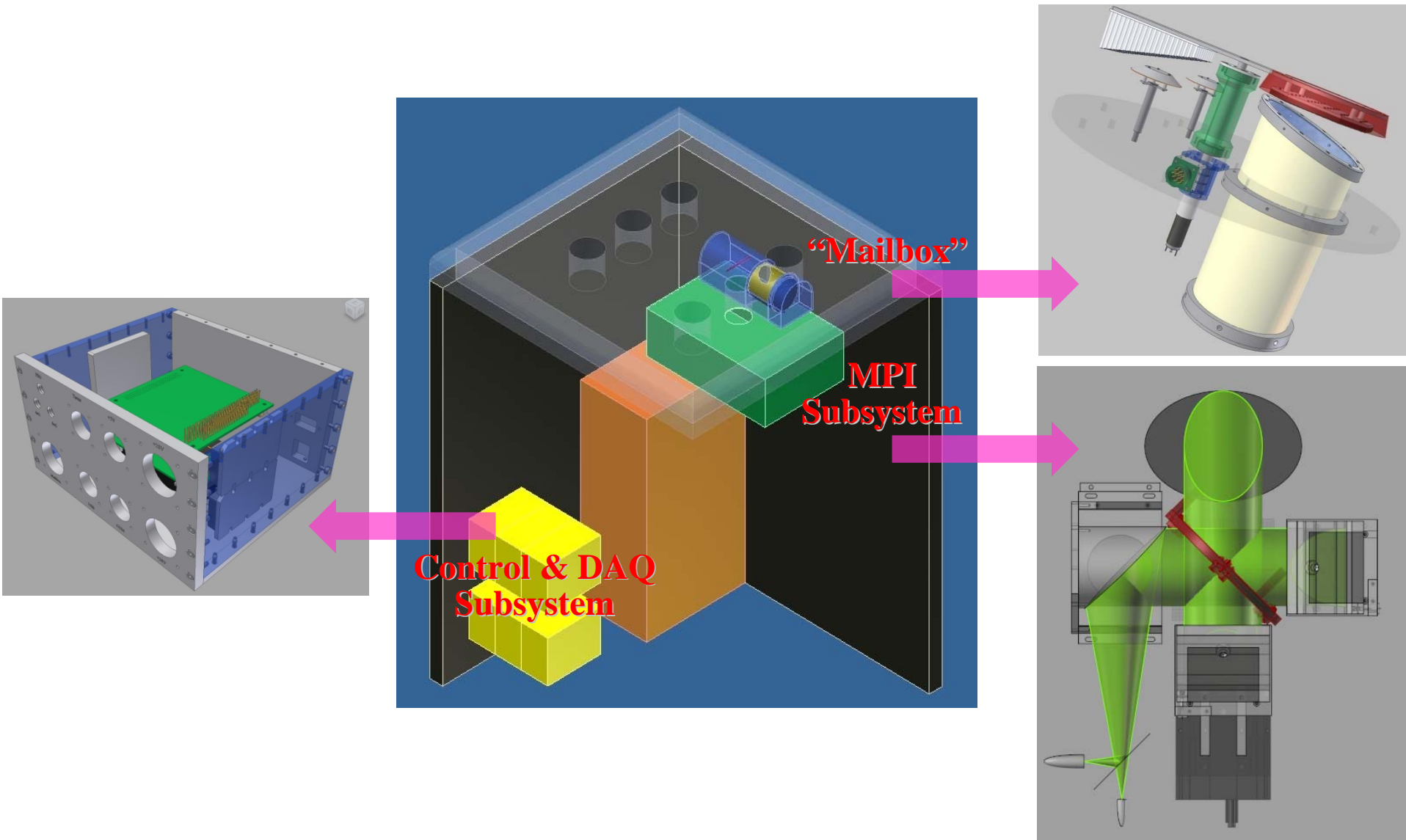
# Design of the FTS System

## — *Basic Requirements*

- unattended operation
- long-duration (~1yr)
- Largest bandwidth for FTSs ever used for site testing
- no LNe used → *no low-temp calibration*
- Cryogenically cooled detectors cannot be used → *limited detection sensitivity*

# Design of the FTS System

## — Schematic of the System





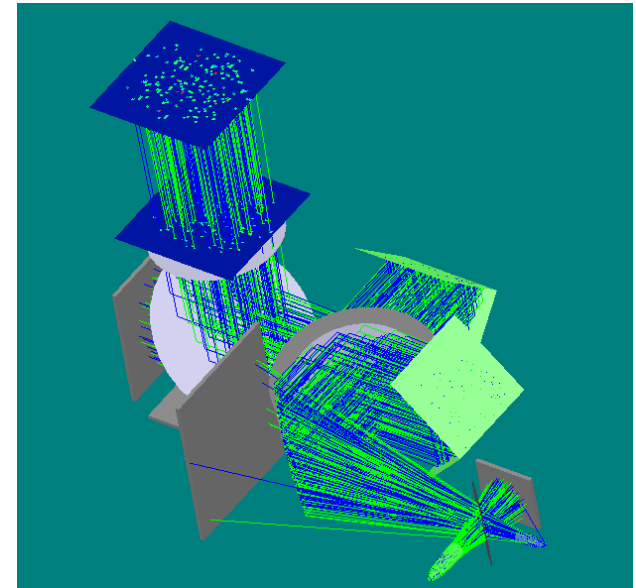
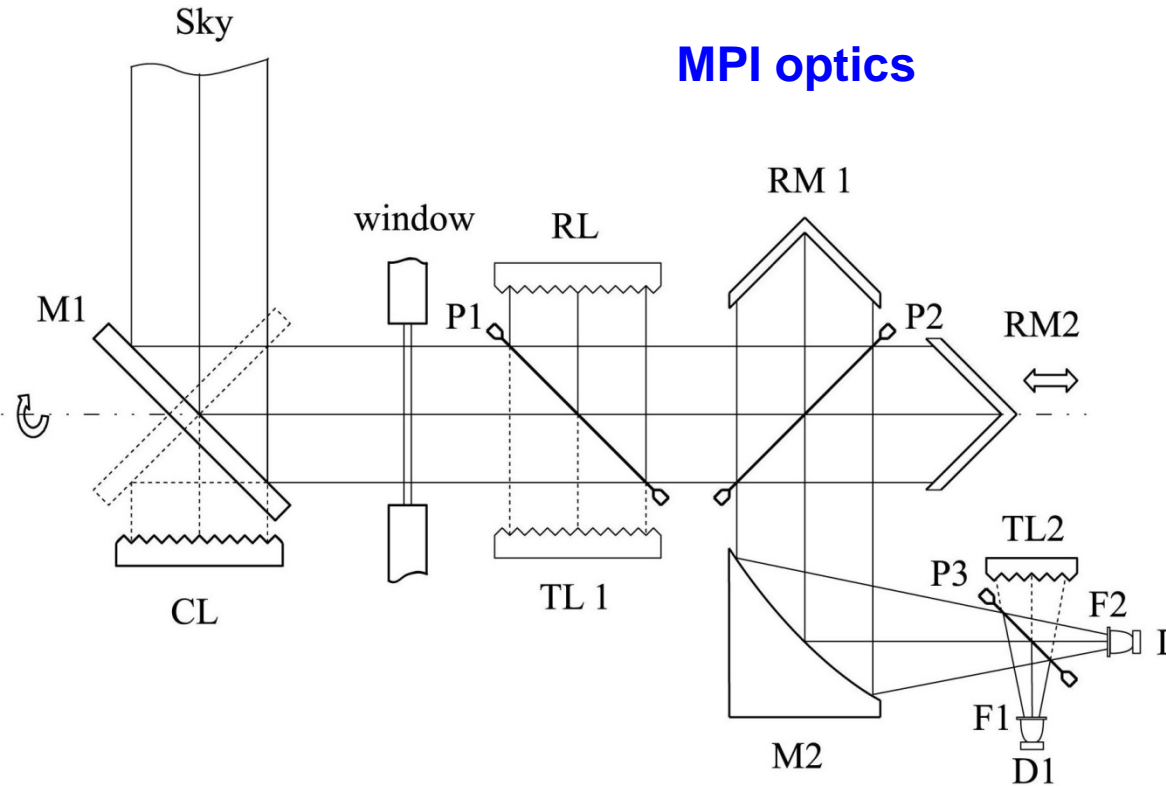
# Design of the FTS System

## — *System Specs*

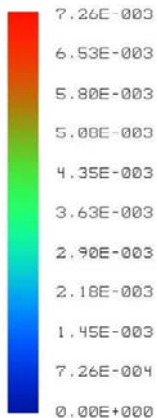
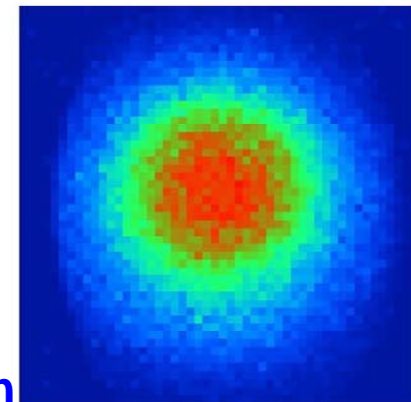
<b>Mode</b>	Fast scan
<b>Freq. Range</b>	0.75-15THz (LB: 0.75-3.5THz, HB: 3.5-15THz)
<b>Freq. Resolution</b>	<10GHz
<b>Beam Aperture</b>	>75mm
<b>DLATGS NEP</b>	$\sim 10^{-10} \text{W/Hz}^{0.5}$
<b>Time/Spectrum</b>	10mins
<b>MPI Volume</b>	0.7mx0.7mx0.3m
<b>Power</b>	<300W

# Design of the FTS System

## — Optics Calculation for MPI

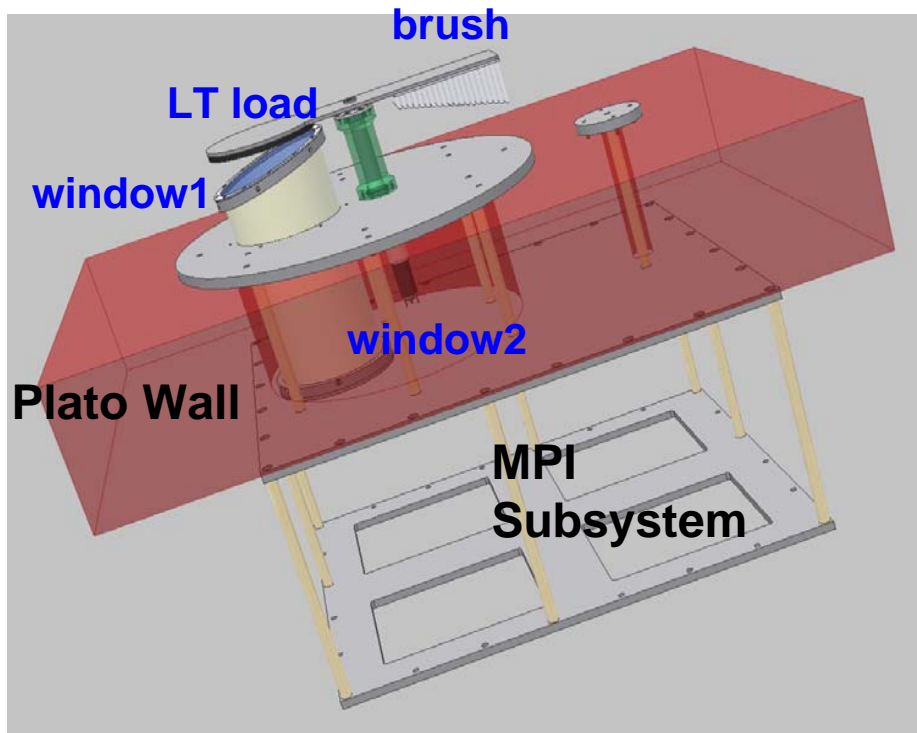


**Zemax  
Simulation**



# Design of the FTS System

## — “Mailbox” Subsystem



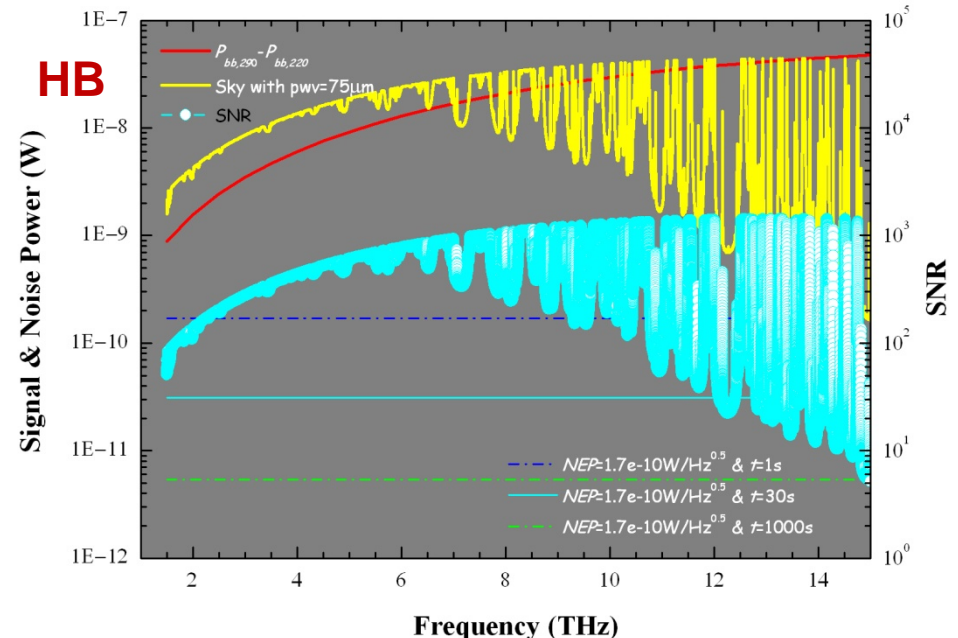
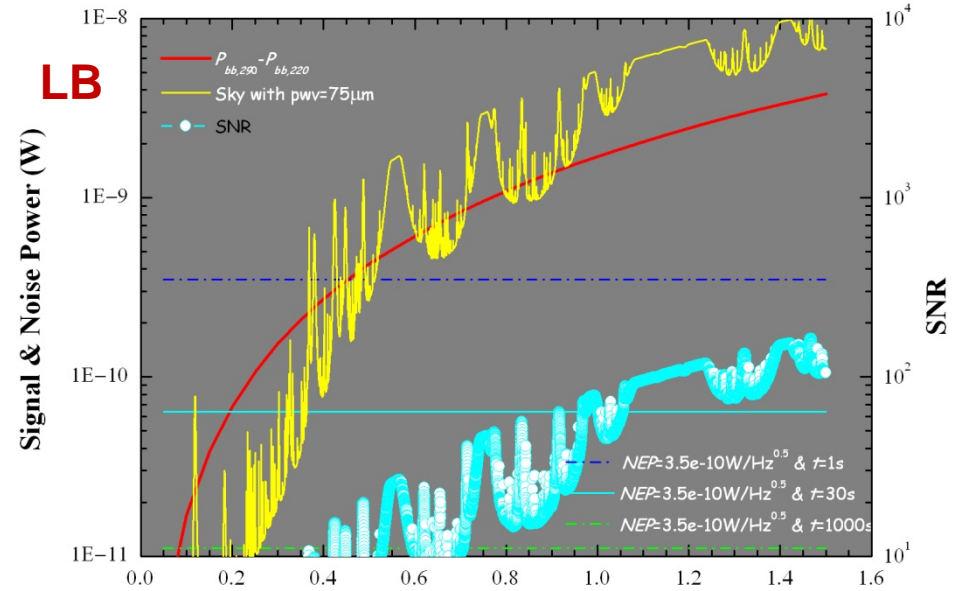
- no tipping for simplicity
- calibration load (LT) & snow cleaning structure integrated
- tilted window plus brush adopted for cleaning snow on the window
- two windows and thermal isolation adopted → less than 20W heat dispersion
- HDPE or LDPE window adopted for large bandwidth
- LT-load radiation temp precisely measured
- driver & oil specially chosen for reliability

# Estimation of Sensitivity

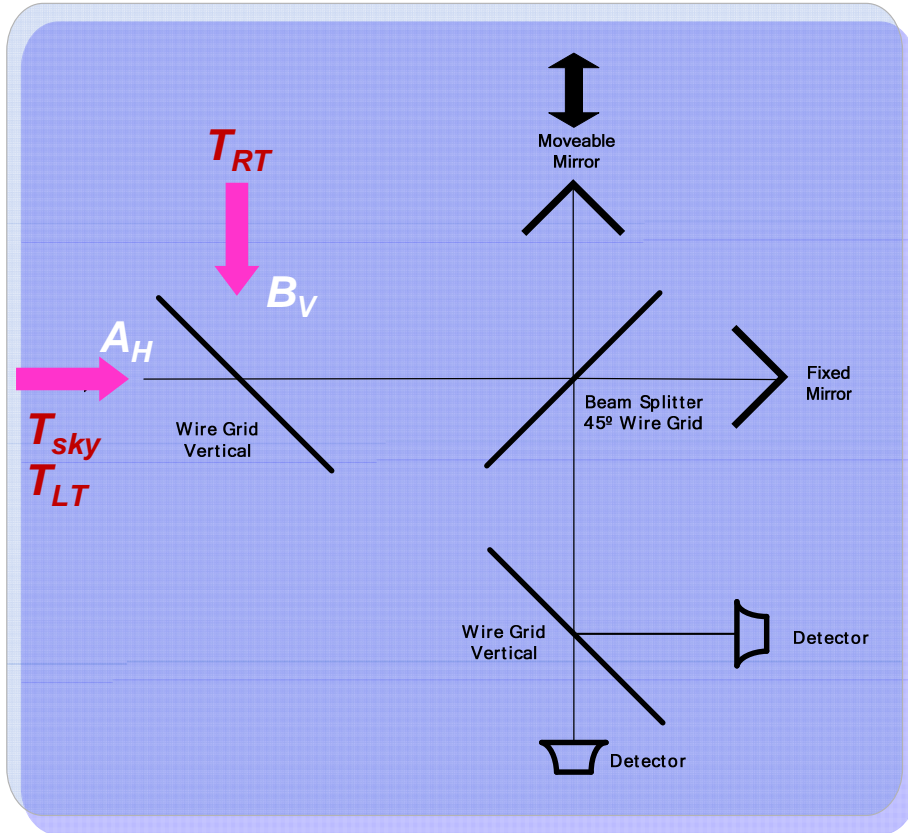
Parameters for Calculation

	<i>Low-band</i>	<i>High-band</i>
<i>Freq Range (THz)</i>	<b>0.5~3.5</b>	<b>3.5~15</b>
<i>Resolution (GHz)</i>	<b>10</b>	<b>10</b>
<i>Optics Efficiency</i>	<b>0.1</b>	<b>0.1</b>
<i>Beam Diameter (cm)</i>	<b>7.5</b>	<b>7.5</b>
<i>Throughput (cm<sup>2</sup>sr)</i>	<b>0.792</b>	<b>0.185</b>
<i>Detector Area (cm<sup>2</sup>)</i>	<b>0.126</b>	<b>0.029</b>
<i>Detector NEP (W/Hz<sup>0.5</sup>)</i>	<b>3.5e-10</b>	<b>1.7e-10</b>

基于DLATGS室温探测器可实现S/N>10



# Scheme of Calibration



$$P_{H,sky-RT}(z, \nu) \propto P_{H,sky}(z, \nu) - P_{V,RT}(\nu) \\ \propto T_{sky,RJ}(z, \nu) - T_{RT,RJ}(\nu)$$

$$P_{H,LT-RT}(\nu) \propto P_{H,LT}(\nu) - P_{V,RT}(\nu) \\ \propto T_{LT,RJ}(\nu) - T_{RT,RJ}(\nu)$$

$z$ : zenith angle,  $\nu$ : frequency

$$\frac{P_{H,sky-RT}(z, \nu)}{P_{H,L-RT}(\nu)} = \frac{T_{sky,RJ}(z, \nu) - T_{RT,RJ}(\nu)}{T_{LT,RJ}(\nu) - T_{RT,RJ}(\nu)}$$

$$T_{sky,RJ}(z, \nu) = \frac{P_{H,sky-RT}(z, \nu)}{P_{H,L-RT}(\nu)} \times \\ [T_{LT,RJ}(\nu) - T_{RT,RJ}(\nu)] + T_{RT,RJ}(\nu)$$

$$S_H(x, \sigma) = \frac{1}{2} \left\{ A_H^2(\sigma) [1 + \cos(2\pi\sigma x)] + B_V^2 [1 - \cos(2\pi\sigma x)] \right\} \\ S_V(x, \sigma) = \frac{1}{2} \left\{ A_H^2(\sigma) [1 - \cos(2\pi\sigma x)] + B_V^2 [1 + \cos(2\pi\sigma x)] \right\}$$

$$S_H(x, \sigma) \xrightarrow{FFT} S_H(\sigma) \propto P_H(\sigma) - P_V(\sigma)$$



# Control/DAQ System & Software

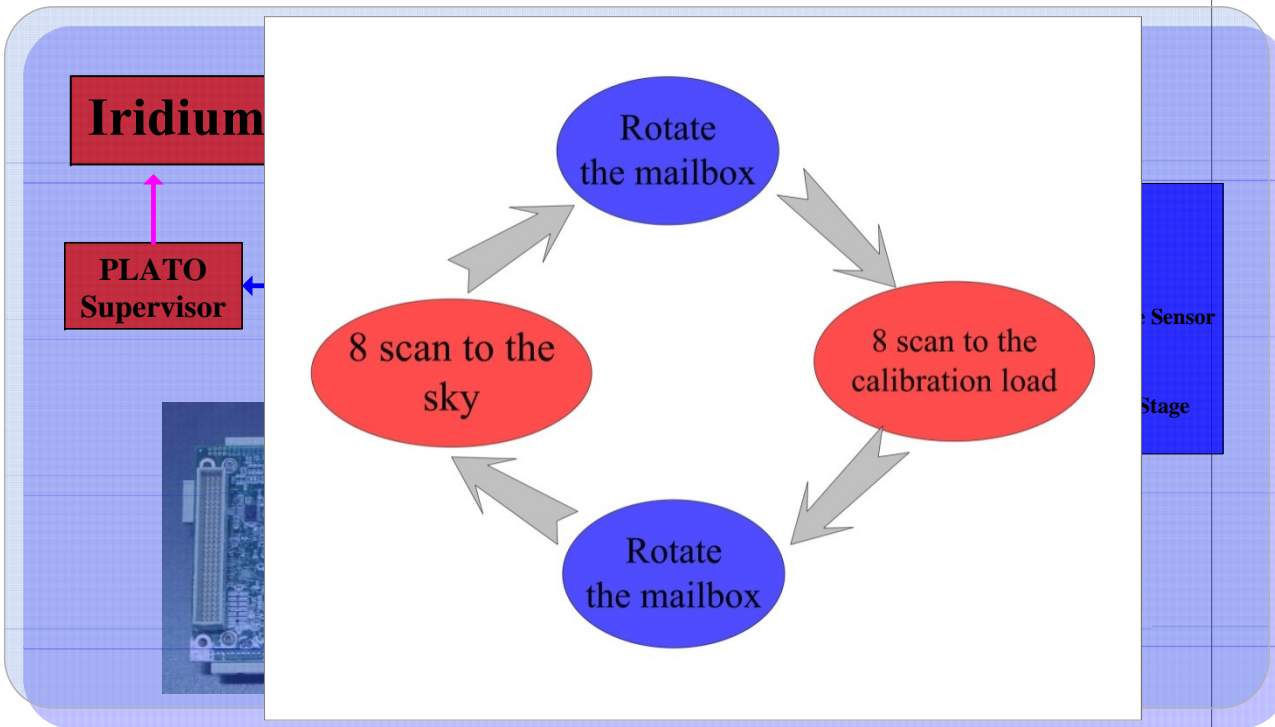
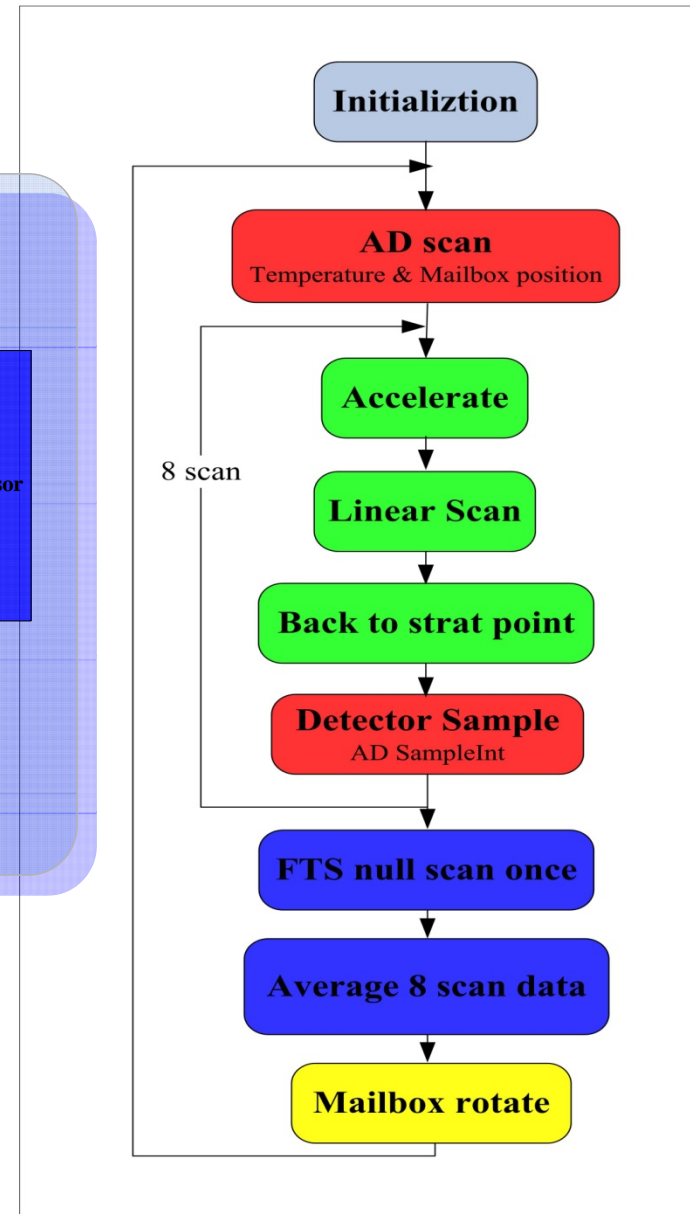


Diagram of Control/DAQ Subsystem



# Timeline for FTS Development

- **End of 2008: Proposal & concept design**
- **2009/1-3: System design and contract negotiation**
- **2009/4: Contracted with QMC/Bluesky for MPI and DLATGS detectors**
- **2009/4-7: Subsystem design & starting construction**
- **2009/8: System integration & lab testing**
- **2009/9: further testing, training, preparing installation/operation manual**
- **2009/10: deploying to expedition team**

# Conclusion

- A THz FTS system for Dome A has been designed,
- Fabrication of subsystems will be completed in July,
- Lab testing and training will start early August,
- Hopefully the FTS will be deployed in time for 2009 Dome A expedition

## Acknowledgment

*Valuable discussions with Ken Wood of QMC, D. Naylor and B. Gom of Bluesky, Plato team, and CCAA team.*